



NEW MILLENNIUM PROGRAM An Overview of NMP Code-Y Activities

Faiza Lansing

March 3, 2000



Purpose of Meeting



On February 4, 2000, F. Li had an informal discussions of with G. Asrar

• Technology validation needs for Earth Science

• Describe updated New Millennium Program structure

Recommended next steps



Challenges to New Millennium Program



- Improve cost-effectiveness of technology flight validation
 - Focus NMP investments on breakthrough technologies/risk reduction
 - Increased reliance on flights-of-opportunities
 - Focus on technology "pieces" requiring flight validation
- Deliver benefits to broad set of Earth Science measurements
 - Reduce cost/enable new capabilities
 - Align technology with science needs
- Enhance partnership for technology development/validation and accelerate technology infusion

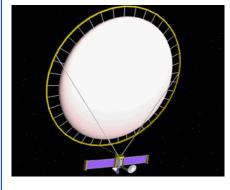


Technology Subsystem Themes

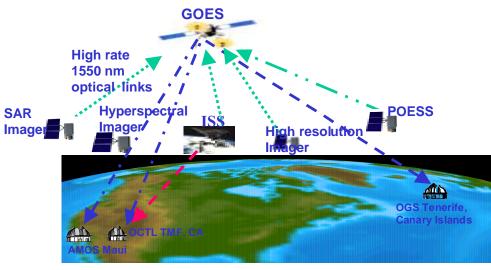


- Preliminary assessment of technology validation needs
- Recurring technology subsystem validation "themes"
- Technology themes benefit a broad set of Earth Science measurements.

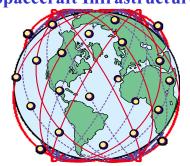
Deployable Light-Weight Microwave/ Milliwave Antennas



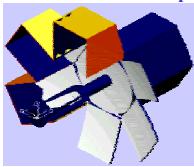
Ultra High Data Rate Communications



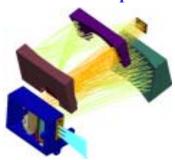
Intelligent Distributed Spacecraft Infrastructure



Light Weight Deployable UV/Visible/IR Telescope



High Performance Spectrometry





Updated New Millennium Program Structure Considerations



Balanced Mix of Breakthrough Subsystem/ Integrated System

- Incorporate testing of technology subsystems
 - Broad benefits to multiple measurement needs
 - Focus on validating the technology "pieces" where needed
- Sharpen criteria for integrated measurement system

Updated NMP Structure

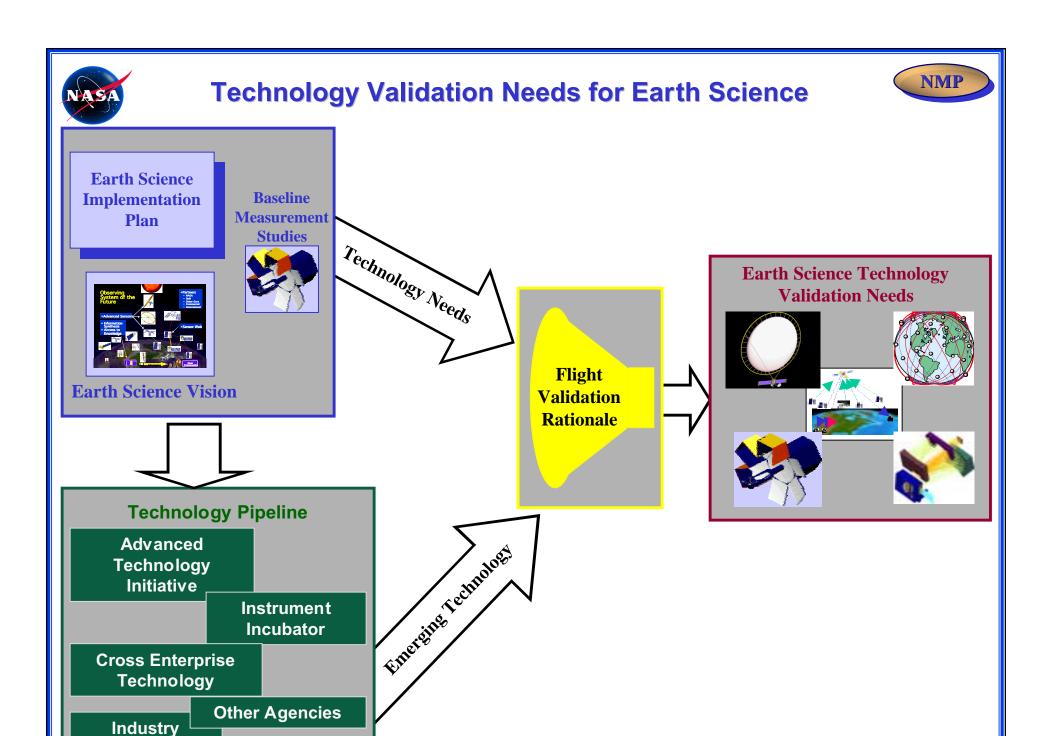
Enhanced Partnership

- Share technology development investment
- Share technology validation platforms
- Improve technology infusion targeting

Stronger Reliance On Flights-of-Opportunity

- Minimize investments in "low tech"
- Broaden validation platform possibilities
 - Including Earth
 science missions

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Technology Validation Needs from Science/Applications Plan

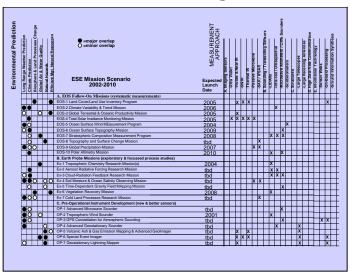


Earth Science Implementation Plan



Applications Strategic Plan

Earth Science Technology Needs Analysis



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Preliminary Soil Moisture Technology Needs

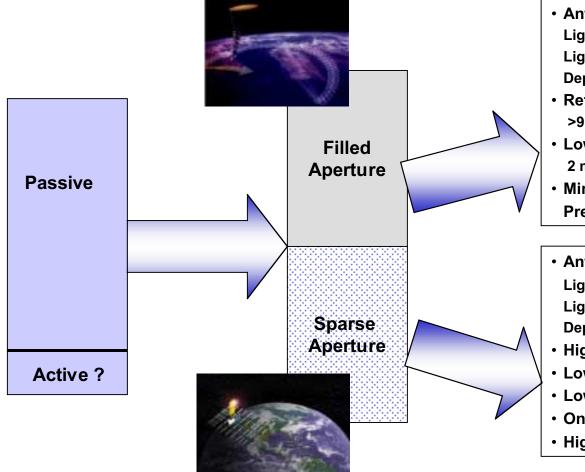


OBJECTIVE: Enable detection of measurement of volumetric moisture in the upper 5-10 cm of the soil with spatial resolution 10-30km and 2-3 day revisit time.

<u>TECHNOLOGY CHALLENGE:</u> Current concepts offer two competing instrument technology paths:

- Synthetic aperture radiometer with thinned antenna array
- Real aperture radiometer with mesh antenna

Both options require investments to answer technological readiness, risk of deployment, structural stability on orbit, and growth potential questions.



Antenna

Lightweight, mesh 6 m , <20 kg Lightweight, 25 m , <200 kg Deployment in zero g

- Reflector surface emission and tolerance
 >90% beam efficiency, <-18dB cross-pol
- Low-mass, low-loss multifrequency feeds
 2 multichannel feedhorns (L/S-band, V/H-pol)
- Miniaturized radiometer/radar electronics
 Precise calibration control

Antenna

Lightweight, inflatable (6m x 10m)<40 kg Lightweight, inflatable (10m x 25m)<150 kg Deployment in zero g

- High performance digital correlators
- Low power, low mass integrated receivers
- · Low loss, high isolation switches
- On-board processing (data compression)
- High rate communications



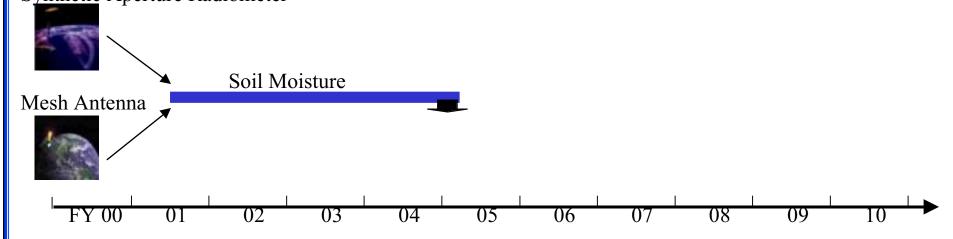
Soil Moisture Measurement: Technology Strategic Pathways



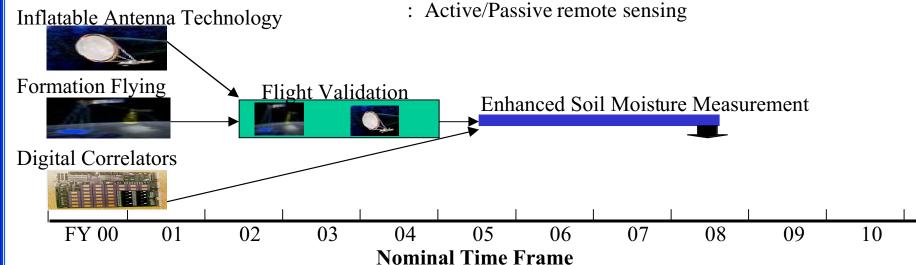
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• Near-term Science Measurement Pathway: 30-40km resolution

Synthetic Aperture Radiometer : Passive remote sensing



• Enhanced Science Measurement Pathway: ~10km resolution





Preliminary Technology Needs for Global Precipitation Measurement Series

Flight

Validations



• Large Aperture Deployable Structure

- 5-meter class inflatable antenna for next generation 14/35 GHz radar or microwave radiometers
- 25-meter class antenna for 35 GHz geostationary radar (mesh/inflatable)

• MMIC Amplifiers

- 50-200 GHz for geosynchronous microwave radiometers/sounders
- 35-GHz T/R Modules & Array Feed
 - Low-loss radar power radiation & electronic scanning

• Autonomous Constellation Control

- Maintain microsat orbital spacings
- Reactive onboard planning (revisit)
- Inter-satellite comms for data handling

• Compact Microwave Radiometers

- Lightweight, low-power for constellation microsats
- 10.7, 37, 85.5 GHz
- Autonomous Space/Ground Internet Protocol

Fine spatial/temporal resolution

Enhanced

Measurements

Latent heat release for climate studies

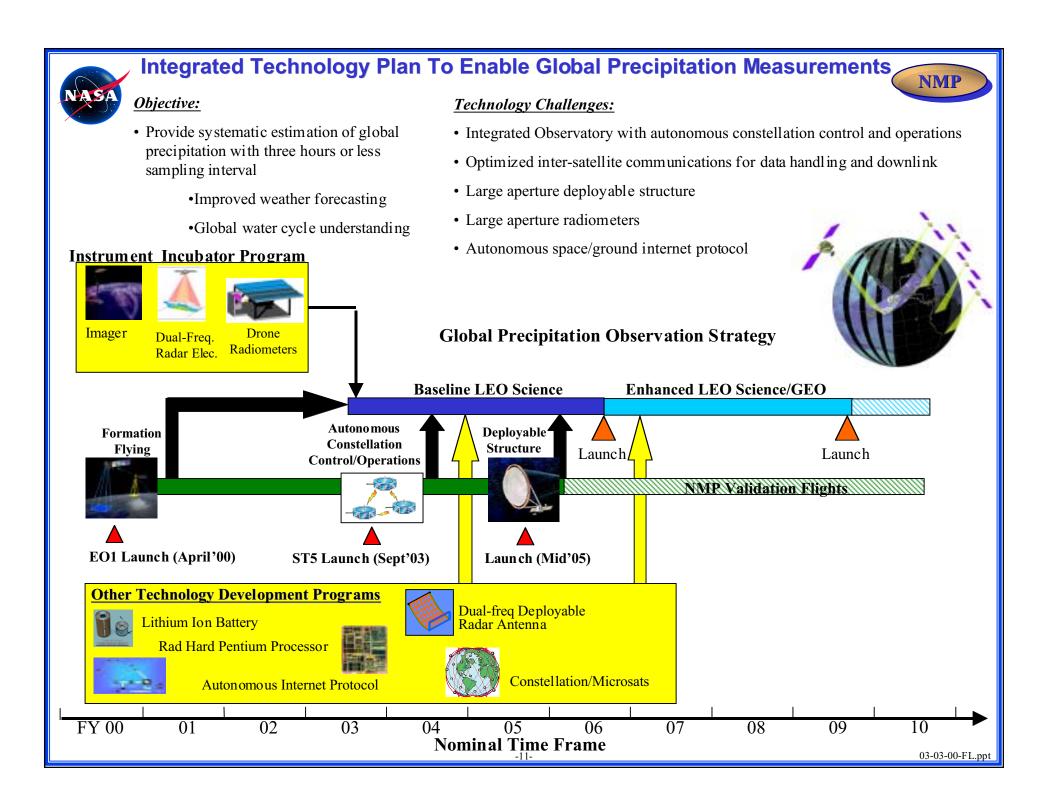
precipitation

patterns

Hurricane/Severe Storm Warning & Landfall Prediction

Baseline Measurements

Rainfall
Climatologies &
Surface Hydrology





Tropospheric Ozone Measurement: Preliminary Technology Needs



UV DIAL: 308/320 nm @ 10Hz; O₃ vertical resolution 2.0-2.5 km in troposphere; horizontal resolution 100 km; IFOV < 100 m Diode-pumped Laser Transmitter (UV) Measurement requirements: 500mJ, 308/320 nm @ 10Hz Current capability: 130mJ, 308/320 nm @ 10Hz **Precision Latch & Hinge** Deployment Mechanisms NMP - flight validation Tropospheric Ozone Measurement Validation Mission Measurement requirement: repeatability reliability, one-time operation Composite **Mirror Panels** Measurement requirement: 3m aperture FY 00 01 02 03 04 05 07 08 09 10 06 Nominal Time Frame -12-03-03-00-FL.ppt



Preliminary Example of Flight Validation Candidate

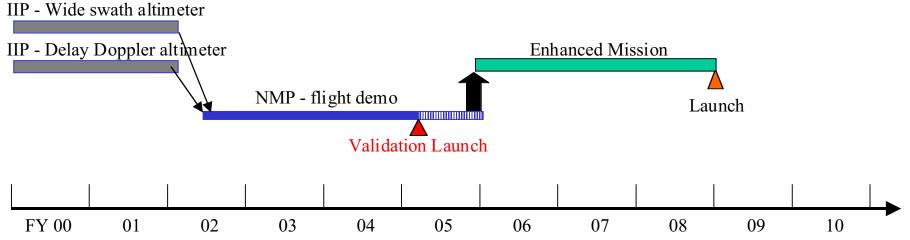


Enhanced Ocean Topography Measurement

• Flight Validation Candidate: Wide swath altimetry

Precision stable structure Delay/Doppler altimeter





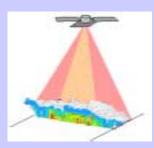
- Flight validation of Integrated Measurement System
 - Ensure critical ocean topography measurement continuity
 - Paradigm shift in measurement approach



Potential IIP Flight Validation Candidates



A Second generation Spaceborne Precipitation Radar (PR-2)



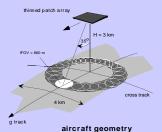
Technology area

5.3 meter dual-frequency(13.6 & 35 Ghz) lightweight (100 Kg) inflatable antenna

Flight Validation Rationale

Test the stability and antenna pattern of a large, light weight inflatable structure for 35 Ghz frequency, 600 KM swath at 2 Km resolution.

Sensing from Space



Technology area

Two Dimensional Synthetic Aperture Radiometer for Microwave Remote

6X10 meter deployable thin array antenna Small digital corrolators

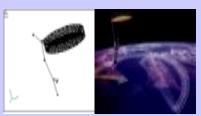
Flight Validation Rationale

Validate the thin array antenna concept Verify structural and thermal stability Verify two-dimensional aperture synthesis concept

Large/lightweight Deployable Antenna

Inflatable Antenna

Spaceborne Microwave Instrument for High Resolution Remote Sensing Using a Large Aperture Mesh Antenna



Technology area

6-meter aperture deployable mesh reflector

Flight Validation Rationale

Validate stability of mesh reflector

Deployable Mesh Antenna

Active Tropospheric Ozone and Moisture Sounder (ATOMS)



Technology area

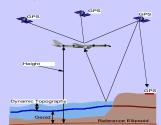
10,22, and 183 GHz links for moisture sounding from 0-~20 km 110 and 165 GHz links for ozone sounding from 8km - ~60 km

Flight Validation Rationale

Validate control infrastructure needed for monitoring, controlling, and orbit maintenance of a constellation of small satellites

Constellation of Small Satellites

GPS-Based Oceanographic and Atmospheric Low Earth Orbiting Sensor (GOALS)



Technology area

Performing surface altimetry using GPS reflections

Flight Validation Rationale

Validate new measurement concept of an on-going measurement

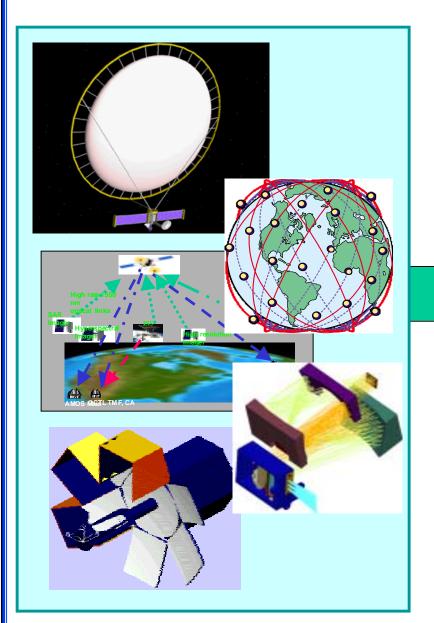
Measurement Technique Using Constellation of Satellites

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Preliminary Subsystem Technology Themes to Support Innovative Earth Science Measurements





- Large Aperture Lightweight Inflatable/Deployable/Optics/Antennas
 - Radiometers
- Radars

Lidars

- Imagers
- Ultra-high Rate Communications/Onboard Processing
 - High spatial/spectral resolution imaging
- Autonomous Constellation Control/Operation
 - Integrated network observatory
 - Coordinated observations at multiple location/vantage points.



Inflatable Microwave/Millimeter Antennas



Component Technologies

Space Inflatable Structures, (TRL 5)

- System architecture
- Deployment control
- Dynamic analysis and simulation
- Scaling laws and ground testing

Rigidization in Space, (TRL 4)

- Low or no power requirements
- Low or no contamination

0 g,

Vacumn & Extreme Temp

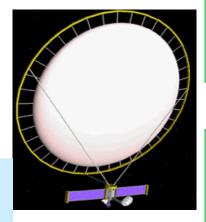
Long-Term Space Survivability, (TRL 4)

- Materials characterization
- Degradation effects of space environment
- >5 year survivability

Membrane Compatible Electronics (TRL 3-4)

- Multi-layer RF membrane microstip array aperture
- (L-band, 80 MHz bandwidth, dual-polarization)
- High frequency membrane reflectorand reflectarrays
- (Ku-band, Ka-band, W-band)
- MEMS T/R Module
- Thin-film solar array

0 g for deployment & performance



Radiation, Atomic O & Micrometeoroid

Measurement Approach

- Altimeter
- Scatterometer
- Synthetic Aperture Radar
- Rain Radar

Science Needs

- Soil Moisture & Ocean Salinity
- Carbon Cycle & Biomass Budget
- Topography & Natural Hazards
- Ocean Surface Wind & Topography
- Land Surface Water & Ice Sheet
- Monitoring
- Global Cloud Mapping &
- Precipitation

Missions

EOS 3,4,6,7, EX 5,6



Ultra-high Data Rate Communications



Component Technologies

Acquisition, Tracking and Pointing

- Acquisition, tracking and pointing algorithms
- 1.55m FPAs
- Laser beacons
- Fast steering mirrors

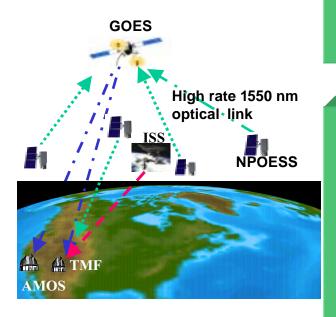
Optical Antennas

• Telescopes, thermally stable 0.1-0.3m aperture

Ultra-High Speed Communications

- Modulators 1-10 Gbps
- Lasers, 1.55m
- Multiplexers and demultiplexers
- Dectectors, 1.55m high speed, low-noise

Precision pointing over long ranges



Supports system level demo from space

Measurement Approach

- Synthetic Aperture Radar
- Hyperspectral Imaging Fourier Transform
- Imaging Spectrometry

Science Needs

High Spatial Resolution:

- Hyperspectral Land Imaging
- Severe Storm Prediction
- Surface Hydrology & Precipitation
- Tectonic Hazard Prediction
- Ozone Monitoring

Missions

EX 6-7, EOS 8, NPOESS, Landsat 9, GIFTS, ...



Intelligent Distributed-Spacecraft Infrastructure

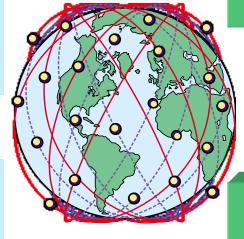


Component Technologies

Formation Flying

- Autonomous reconfiguration
- Virtual dynamical coupling
- Collision avoidance
- Optimal fuel strategy
- Orbit maintenance
- System modeling and coordination

Highprecision measurements with large number of stand-alone instruments



Measurement Approach

- Multiple Networked Orbiters
- Sensor Webs
- Smart Sensors

Sensor/Satellite Networks

- Continuous communication coverage via asymmetric, hybrid links
- Management of a complex, multinode heterogeneous network
- Scalable design for incremental network growth
- Graceful degradation to network performance during stress
- Robust routing, adaptive bandwith allocation, and intelligent power control of nodes

Globalcoverage
scalable
network
for robust,
ondemand,
highthroughout
data
transfer

Science Needs

- Atmospheric Chemistry
- Temperature, Clouds & Radiation Budget
- Global Precipitation
- Hazard Monitoring
- Diurnal Vegetation Monitoring

Missions

EOS 1,3,5,6,7,9 OP 3,6

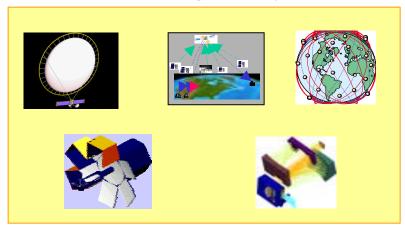


Augment NMP Program Content with Enabling Breakthrough Subsystem Perspective



• Balanced mix of of subsystem/integrated measurement systems

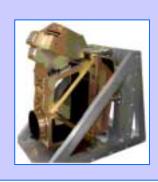
Breakthrough Subsystems



- Breakthrough subsystems that
 - Require flight validation (environment, paradigm shift)
 - Enable critical functions for key/enhanced measurements
 - Yield broad benefits to multiple measurement systems
- Breakthrough subsystems can be tested as stand-alone items without full instruments
 - More cost effective
 - Focus on validating technologies where needed

Integrated Measurement System





- Ensure critical science measurement continuity
- Required risk reduction for transition to operational measurements
 - Fundamental paradigm shift in measurement approach

Sharpen Current NMP Criteria

Augmentation to include validation of breakthrough subsystems

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Updated New Millennium Program Attributes



Integrated System/ Subsystem Mix

- Incorporate subsystem test path
 - Breakthrough subsystems requiring flight validation
 - Broad benefits to multiple measurement systems
 - Enable critical functions
- Sharpen integrated measurement system criteria
- Balanced mix of integrated/subsystem flights

Strategic Mix of Focused/Broad Opportunities

- Focused technology validation opportunities to address strategic needs
- Broad opportunities to solicit innovative ideas
- Technologies always acquired via open, peerreviewed competitions

Updated NMP Structure Attributes Strong reliance on Flights-of-Opportunity

- Improve cost-effectiveness
- "Reservation" on Earth Science missions for technology validation
- Broaden validation platform possibilities (shuttle, station, other agencies, commercial, etc.)

Partnership for Technology Investments

- Partnership in
 - Validation platform
 - Technology investment
- Potential Partners
 - Other NASA Codes
 - Non-NASA agencies
 - Industry

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Summary/Recommended Next Steps



- Initiated identification of technology validation needs for Earth Science
- Examined updated program structure
 - Improve cost effectiveness
 - Mix of breakthrough subsystem/innovative measurement system
 - Enhanced partnership
- Recommended next steps
 - Assess breakthrough subsystem validation requirements/approaches
 - ♦ Conduct mini-workshops on technology validation need (3 or 4 focused areas)
 - Status review/feedback sessions with YS/YO in late March'00
 - Presentation in April '00 to Code Y/AA
 - Synthesize technology investment advocacy package



DRAFT NMP 2000 Workshops - GENERAL



- One day workshop
 - A. Large, light weight Microwave/Millimeter Antennas
 - B. Ultra-High Data Rate Communication
 - C. Intelligent Distributed Spacecraft Infrastructure
 - D. High Performance Spectrometry
 - E. Light Weight Deployable UV/Visible/IR Telescope
- Purpose: To develop technology validation experiment/development plan for specific technology areas to address OES science and application needs
- **Discuss:** Technology requirements from science measurements perspective
 - Technology development plan (5-10 yr. planning horizon)
 - What is the right space validation experiment
- Will invite, with HQ concurrence for each workshop
 - 5 Scientists*
 - 5 Technologists
 - 2 Co-Chairs (Science & Technology)
 - 1 Facilitator
 - * Code YS Discipline Scientists to be invited to each workshop
- Week of April 3 7 or before (goal of three workshops completed this week, balance by 4/19)



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DRAFTNMP 2000 Workshops



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Large, Light-Weight Deployable Antennas

Co-Chairs

- Science: T. England (U of M)

- Technology: C. Moore (LaRC)

- Workshop Facilitator: D. Crisp (NMP)

Location: JPL, Date: 4/13/00

Ultra-High Data Communication В.

Co-Chairs

- Science: A. Goetz (U of C)

- Technology: K. Bhasin (Glen)

- Workshop Facilitator: F. Lansing (NMP)

Location: GSFC, Date: 4/7/00

Intelligent Distributed - Spacecraft Infrastructure

Co-Chairs

- Science: G. North

- Technology: W. Kaiser (UCLA)

- Workshop Facilitator: C. Raymond (NMP)

Location: GSFC, Date: 4/6-8/00

Scientists Technologists

John Huang

Michael Lou

Scientists Technologists

Martha Maiden Hamid Hemmati Tim Berkoff Godfrey Anzic? Dennis Andrucyk James Dodge

Rob Green **Phil Luers**

Greg Prescott? Bob Murphy

Scientists Technologists



DRAFT NMP 2000 Workshops



D. High Performance Spectrometry

Co-Chairs Dar Roberts (UCSB)

- Science: D. Wickland Roger Clark (USGS)

- Technology: TBD Joe Boardman (AIG)

Workshop Facilitator: C. Stevens

Laurie Richardson (FIU)

Location: JPL Susan Ustin (UCD)

Greg Asner (Uof C)

E. Light Weight Deployable UV/Visible/IR Telescope

Co-Chairs

- Science: TBD

- Technology: F. Peri

Workshop Facilitator: V. Sarohia

Location: LaRC





TITLE OF TECHNOLOGY/BREAKTHROUGH CAPABILITY

Description of Technology

Technology Picture(s)

Relevance/Importance to Future Code Y Missions

Measurements & Requirements "View of the Scientists"





TITLE OF TECHNOLOGY/BREAKTHROUGH CAPABILITY

Description of the SOA of the Technology

Picture Showing Current Technology Products and/or Ground Test Models

Major Technology Elements and Current TRL Technology Development Roadmap Showing Space Validation Flight(s) as Key Milestones





TITLE OF TECHNOLOGY/BREAKTHROUGH CAPABILITY

Description of the Proposed Space Validation and Justification for Validation in Space Expected Benefits, Including
Enabled Measurements and
Missions, Enabled Critical
Functionality, and/or Expected
Life-Cycle Cost Reduction

Projected Cost and WF Requirements
By FY

Top-Level Development and Flight Schedule





THE SPACE VALIDATION EXPERIMENT(S)

What is the "Right" Space Experiment (Consensus of the Science and Technology Communities)

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THE NMP PERFORMANCE METRICS



- ★ Each project selected for formulation shall respond to at least one technology requirement for more than one mission in the Earth Science or Space Science Enterprise Strategic Plan.
- The selection of the destination for each flight validation project shall be determined based upon the requirements for technology validation and not be driven by requirements for science data collection.
- Technologies eligible for selection for a new project shall have completed technology development to a technology readiness level where the concept design has been tested experimentally.
- Technologies selected for a project shall have completed testing of the pre-prototype prior to project approval and initiating project implementation
- **★** Over a five-year running time period, a minimum of 80 percent of the approved project-defining technologies shall be flight validated, and 100 percent of the project-defining technologies that are launched in a project shall be flight validated.
- ★ Over a five-year running time period, a minimum of 70 percent of the approved project-enhancing technologies shall be flight validated.
- Advances in technology development that occur prior to launch shall be documented in annual updates to the NASA Technology Inventory and in a report prepared and submitted by the NMP Office to the applicable Enterprise as part of the annual budget review.
- ★ Data from technology flight validations shall be documented and made available to technology providers and mission planners within six months of the completion of flight validation subject to the restrictions imposed by Export Administration Regulations and International Traffic in Arms Regulations.

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